# 16.513 Control systems (Lecture #11)

Last time,

- Controllability and observability (Chapter 6)
- Two approaches to state feedback design (Chapter 8)
  - Using controllable canonical form
  - By solving matrix equations

Today, we continue to work on feedback design (Chapter 8)

- Other state-feedback problems
  - Regulation and tracking
  - Robust tracking and disturbance rejection
  - Stabilization of uncontrollable systems
- Full dimensional estimator
  - SISO case via observable canonical form
  - MIMO case by solving matrix equation

1

### Project and Final Exam: Due 12pm, Dec 16, 2018

Final exam problems will be sent to your email box at uml, at 9am, Dec 15 (Saturday).

- The written part of the project should be complete with all results clearly presented. 3 points out of 25 will be given on presentation.
- All the Matlab and Simulink files for the project and the final exam should be contained in a zip file for possible verification.
- The project and final exam should be done independently.

Please send 4 files to me via email between 12 - 12:30pm, 12/16/18

3

- 1) Project; 2) Final exam; 3) Homework #12 (pdf file, or MS word)
- 4) Zip file for all Matlab/Simulink source files

**Regulation and Tracking** (§8.3)

- Generally, regulation is about bringing the output or state to certain desired value asymptotically and keep it there. It can be transformed into a stabilization problem
- Tracking is a relationship between the output and reference signals. It describes the property of how the output y(t) follows a desired reference r(t).
  - The simplest tracking problem is to track a step signal.
  - A more complicated case is to track a sinusoidal signal, a polynomial signal, or periodic signals.
  - We need more advanced tools to address the second case.

#### Tracking a step signal

Recall that we use u = r - kx to stabilize a system and the resulting closed-loop system is

 $\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{b}\mathbf{k})\mathbf{x} + \mathbf{b}\mathbf{r}, \quad \mathbf{y} = \mathbf{c}\mathbf{x}$ 

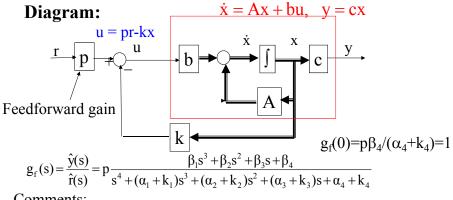
By choosing k appropriately, the transfer function from r to y is

$$g(s) = c(sI - A + bk)^{-1}b = \frac{\beta_1 s^3 + \beta_2 s^2 + \beta_3 s + \beta_4}{s^4 + (\alpha_1 + k_1)s^3 + (\alpha_2 + k_2)s^2 + (\alpha_3 + k_3)s + \alpha_4 + k_4}$$

Suppose that  $\beta_4 \neq 0$ . Then the DC gain from r to y is  $g(0)=\beta_4/(\alpha_4 + k_4)$ . If g(0) is 1 and r(t) is a step, then  $y(t)-r(t) \rightarrow 0$ . If  $g(0)\neq 1$ , we need to introduce a feedforward gain p, i.e., let u = pr-kx, with  $p=(\alpha_4+k_4)/\beta_4$ . Then

$$g_{f}(s) = \frac{\hat{y}(s)}{\hat{r}(s)} = p \frac{\beta_{1}s^{3} + \beta_{2}s^{2} + \beta_{3}s + \beta_{4}}{s^{4} + (\alpha_{1} + k_{1})s^{3} + (\alpha_{2} + k_{2})s^{2} + (\alpha_{3} + k_{3})s + \alpha_{4} + k_{4}}$$

$$g_{f}(0) = 1 \implies \text{The output y(t) can track any step signal r(t).} \qquad 5$$



Comments:

- The feedforward strategy should work well when the parameters are accurate and there is no external disturbance.
- However, if the parameters have errors, p≠(α<sub>4</sub>+k<sub>4</sub>)/β<sub>4</sub>, then the final DC gain may not be exactly one. Also, some disturbance may cause steady-state tracking errors.
- Robust tracking problem is formulated to deal with these issues.

#### **Robust Tracking and disturbance rejection**

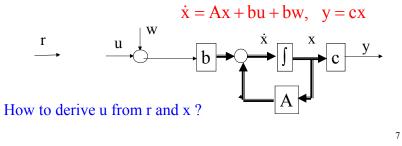
Consider an open-loop system

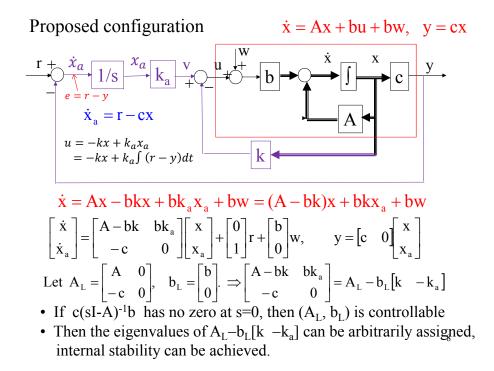
 $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{b}\mathbf{u} + \mathbf{b}\mathbf{w}, \quad \mathbf{y} = \mathbf{c}\mathbf{x}$ 

where w is the disturbance. Assume that (A,b) is controllable. Suppose that there are uncertainties in A, b and c:

 $A \rightarrow A + \delta A, b \rightarrow b + \delta b, c \rightarrow c + \delta c.$ 

Robust tracking requires y(t) to follow a step r(t) in the presence of uncertainties and disturbances.





• Now we discuss tracking and disturbance rejection:

Let 
$$\Delta_{f}(s) = \det(sI - A_{L} + b_{L}[k - k_{a}]) = \det\begin{bmatrix}sI - A + bk & -bk_{a}\\c & s\end{bmatrix}$$
  
Suppose  $c(sI - A + bk)^{-1}b = N(s)/D(s)$  Then  $\Delta_{f}(s) = sD(s) + k_{a}N(s)$   
Let  $y(s) = y_{r}(s) + y_{w}(s) = \hat{g}_{r}(s)r(s) + \hat{g}_{w}(s)w(s)$   
 $\hat{g}_{r}(s) = \frac{y_{r}(s)}{r(s)} = \frac{k_{a}N(s)}{\Delta_{f}(s)} = \frac{k_{a}N(s)}{sD(s) + k_{a}N(s)} \implies \hat{g}_{r}(0) = 1$   
 $\hat{g}_{w}(s) = \frac{sN(s)}{\Delta_{f}(s)}, \quad \hat{g}_{w}(0) = 0$  If w(t) is a step signal,  $y_{w}(t) \rightarrow 0$ 

Conclusions:

- The DC gain from r to y is always 1. Tracking step signal asymptotically even if parameters A,b,c change.
- The DC gain from w to y is always 0. Step disturbance can be rejected.

9

Example: 
$$u = g_o(s) = \frac{s^2 + 2s + 3}{s^3 + s^2 + s - 1}$$
 y

Design a robust tracking control strategy such that y tracks a step signal r(t) asymptotically.

State space realization of  $g_o(s)$ :

$$A = \begin{bmatrix} -1 & -1 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}, \quad C = \begin{bmatrix} 1 & 2 & 3 \end{bmatrix}$$
$$A_{L} = \begin{bmatrix} A & 0 \\ -c & 0 \end{bmatrix} = \begin{bmatrix} -1 & -1 & 1 & 0 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ -1 & -2 & -3 & 0 \end{bmatrix}, \quad b_{L} = \begin{bmatrix} b \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix},$$

The basic requirement is stability. The convergence rate depends on the eigenvalues of  $A_L-b_Lk_L$  ( $k_L=[k - k_a]$ ) We first choose  $k_L$  that assign the eigenvalues. 10

We use the second approach of pole assignment. Pick

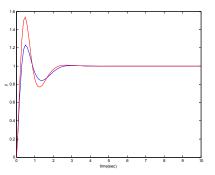
$F = \begin{bmatrix} -2 & -2 & 0 & 0\\ 2 & -2 & 0 & 0\\ 0 & 0 & -4 & -4\\ 0 & 0 & 4 & -4 \end{bmatrix}$	Pick k0=[1 1 1 1]; T=lyap(AL,-F,-bL*k0) k <sub>L</sub> =k0*inv(T)
$k_L = [11.0000 - 14.3333]$ k=[11.0000 - 14.3333]	

y(t) is plotted in red curve.

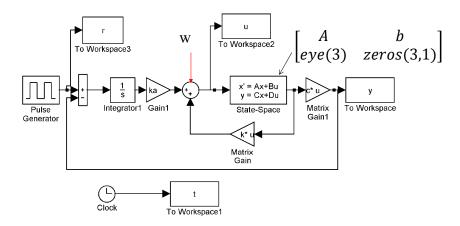
If we pick

$$F = \begin{bmatrix} -2 & -2 & 0 & 0\\ 2 & -2 & 0 & 0\\ 0 & 0 & -4 & 0\\ 0 & 0 & 0 & -8 \end{bmatrix}$$

The tracking performance is improved. See y(t) plotted in blue.

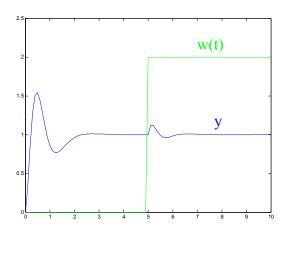


The simulink model

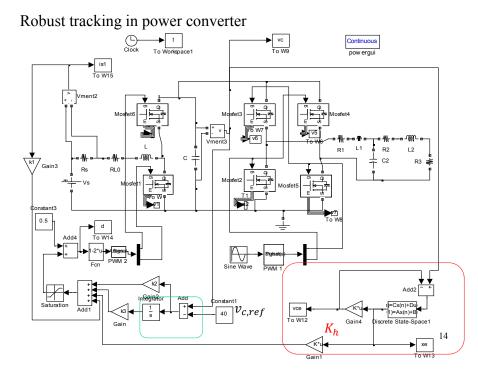


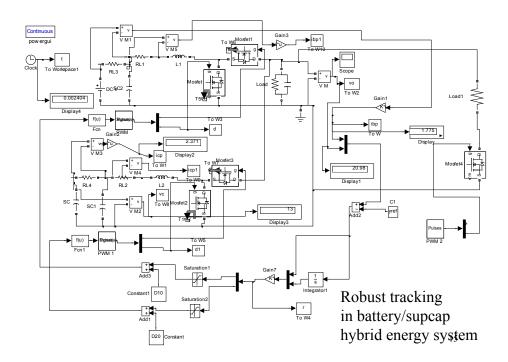
You can try to change the parameters A, or b and the tracking property is maintained.

A response in the presence of step disturbance









#### Problem set #11

1. Design a robust tracking strategy for the system

$$u \rightarrow g_o(s) = \frac{s^2 + s + 1}{s^3 - s^2 + 2s - 1}$$
   
  $y \rightarrow y$ 

so that the output y follows a step signal asymptotically. Choose design parameters so that the closed-loop poles are at -2+j2, -2-j2, -4 and -8. Simulate the system from t=0s to t=20s. (print the simulink model) 1) with the given  $g_0(s)$ 

2) Keep all the design parameters but replace  $g_0(s)$  with

$$g_1(s) = \frac{s^2 + (1+\delta)s + 1}{s^3 - s^2 + (2-\delta)s - 1}$$
 for  $\delta = 0, 0.5, 1, 2$ 

Plot y(t) for each of the cases with 0 initial condition for the state. You may plot all responses in the same figure and identify them with the value of  $\delta$ .

What is the minimal  $\delta$  to make the closed-loop system unstable? (Note: use the same state-feedback for all cases.)

### Stabilization of uncontrollable systems

Recall: If (A,B) is controllable, then the eigenvalues of (A+BK) can be arbitrarily assigned.

• What if (A,B) is not controllable? Can the system be stabilized?

The original system:  $\dot{x} = Ax + Bu$ 

Suppose that the system is transformed (by z = Px) into the following:

$$\dot{z} = \overline{A}z + \overline{B}u, \quad \overline{A} = PAP^{-1}, \quad \overline{B} = PB, \quad z = \begin{bmatrix} z_1 \\ z_2 \end{bmatrix}$$
$$\begin{bmatrix} \dot{z}_1 \\ \dot{z}_2 \end{bmatrix} = \begin{bmatrix} \overline{A}_c & \overline{A}_{12} \\ 0 & \overline{A}_{\overline{c}} \end{bmatrix} \begin{bmatrix} z_1 \\ z_2 \end{bmatrix} + \begin{bmatrix} \overline{B}_c \\ 0 \end{bmatrix} u, \quad (\overline{A}_c, \overline{B}_c) \text{ controllable}$$

Note:  $\operatorname{eig}(A) = \operatorname{eig}(\overline{A}) = \operatorname{eig}(\overline{A}_c) \cup \operatorname{eig}(\overline{A}_{\overline{c}})$ 

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$$\begin{bmatrix} \dot{z}_{1} \\ \dot{z}_{2} \end{bmatrix} = \begin{bmatrix} \overline{A}_{c} & \overline{A}_{12} \\ 0 & \overline{A}_{c} \end{bmatrix} \begin{bmatrix} z_{1} \\ z_{2} \end{bmatrix} + \begin{bmatrix} \overline{B}_{c} \\ 0 \end{bmatrix} u, \quad (\overline{A}_{c}, \overline{B}_{c}) \text{ controllable}$$
Note:  $\operatorname{eig}(A) = \operatorname{eig}(\overline{A}) = \operatorname{eig}(\overline{A}_{c}) \cup \operatorname{eig}(\overline{A}_{c})$ 
Consider a state feedback:  $u = r \cdot Kx = r \cdot \overline{K}z = r \cdot [\overline{K}_{1} \quad \overline{K}_{2} \begin{bmatrix} z_{1} \\ z_{2} \end{bmatrix}, \quad K = \overline{K}P$ 

$$\implies \begin{bmatrix} \dot{z}_{1} \\ \dot{z}_{2} \end{bmatrix} = \begin{bmatrix} \overline{A}_{c} - \overline{B}_{c}\overline{K}_{1} & \overline{A}_{12} - \overline{B}_{c}\overline{K}_{2} \\ 0 & \overline{A}_{c} \end{bmatrix} \begin{bmatrix} z_{1} \\ z_{2} \end{bmatrix} + \begin{bmatrix} \overline{B}_{c} \\ 0 \end{bmatrix} r,$$
Then  $\operatorname{eig}(A \cdot BK) = \operatorname{eig}(\overline{A} \cdot \overline{B}\overline{K}) = \operatorname{eig}(\overline{A}_{c} - \overline{B}_{c}\overline{K}_{1}) \cup \operatorname{eig}(\overline{A}_{c})$ 
Conclusion:
$$= \operatorname{Since}(\overline{A} \quad \overline{B}) \text{ is controllable } \operatorname{eig}(\overline{A} - \overline{B} \mid \overline{K}) \text{ can be}$$

- Since  $(A_c, B_c)$  is controllable,  $eig(A_c B_cK_1)$  can be arbitrarily assigned
- $eig(\overline{A}_{\overline{c}})$  cannot be changed by any state feedback. They are called uncontrollable modes.
- For the system to be stabilizable, the uncontrollable modes have to be stable, i.e.,  $\operatorname{Re}(\lambda_i(\overline{A}_{\overline{c}})) < 0$  for each i.

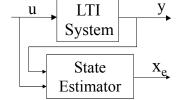
Today's topics:

- Other state-feedback problems
  - Regulation and tracking
  - Robust tracking and disturbance rejection
  - Stabilization of uncontrollable systems
- Full dimensional estimator
  - SISO case via observable canonical form
  - MIMO case by solving matrix equation

#### 19

# **State Estimators**

- Previously, we assumed that x is available. However, generally x is not available
- **Q.** What to do?
  - Construct a system to estimate x ~ State estimator (or observer)



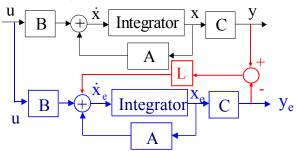
- Want x<sub>e</sub>(t) to be a "good" estimate of x(t), i.e.,
  - x(t) x<sub>e</sub>(t) should go to zero asymptotically with fast convergence rate.

- Main result: If the system is observable, then by properly designing the state estimator, the poles of the dynamics associated with  $x(t) x_e(t)$  can be arbitrarily assigned
  - > Error goes to zero as quickly as possible
- We will discuss several types of estimators
  - Full-dimensional state estimator
    - Single output case
    - Multivariable output case
  - Reduced-dimensional state estimator
- Finally, connecting state estimator and state feedback

21

## **Full-Dimensional State Estimators**

- How to estimate x based on u and y?  $\dot{x} = Ax + Bu; y = Cx \sim Assuming that D = 0$
- A proposed configuration:



- Duplicate the system dynamics
- Make correction on  $dx_e/dt$  when y and  $y_e$  are different
- Pick the correction as a linear function of  $(y y_e)_{22}$

## Will the proposed configuration work?

The original system

 $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}; \quad \mathbf{y} = \mathbf{C}\mathbf{x}$ 

The duplicated system (estimator)

 $\dot{\mathbf{x}}_e = \mathbf{A}\mathbf{x}_e + \mathbf{B}\mathbf{u} + \mathbf{L}(\mathbf{y} - \mathbf{y}_e), \quad \mathbf{y}_e = \mathbf{C}\mathbf{x}_e$ 

- The correction term  $L(y-y_e)$  plays the essential role.
- Can we choose L appropriately to make x<sub>e</sub> approach x?

- We now analyze the system
  - First, consider the estimator dynamics

$$\dot{\mathbf{x}}_{e} = \mathbf{A}\mathbf{x}_{e} + \mathbf{B}\mathbf{u} + \mathbf{L}(\mathbf{y} - \mathbf{y}_{e}), \qquad \mathbf{y}_{e} = \mathbf{C}\mathbf{x}_{e}$$
$$= \mathbf{A}\mathbf{x}_{e} + \mathbf{B}\mathbf{u} + \mathbf{L}[\mathbf{C}\mathbf{x} - \mathbf{C}\mathbf{x}_{e}]$$
$$= (\mathbf{A} - \mathbf{L}\mathbf{C})\mathbf{x}_{e} + \mathbf{L}\mathbf{C}\mathbf{x} + \mathbf{B}\mathbf{u}$$

- Next, consider the dynamics of  $e:=x x_e$   $\dot{e} = \dot{x} - \dot{x}_e = (Ax + Bu) - [(A - LC)x_e + LCx + Bu]$   $= (A - LC)(x - x_e) = (A - LC)e$  $\dot{e} = (A - LC)e$
- If the eigenvalues of (A LC) have negative real parts, any error will converge to 0, x - x<sub>e</sub> → 0.
- Can the eigenvalues of A-LC be arbitrarily assigned? 24

**Theorem.** If the system (A,C) is observable, then all the eigenvalues of (A - LC) can be arbitrarily placed, provided that complex eigenvalues appear in pairs

**Proof** : The result follows from duality

- The eigenvalues of (A-LC) and (A'-C'L') are the same.
- By Theorem 6.5: (A, C) is observable iff (A', C') is controllable (page 155)
- Now, since (A, C) is observable, (A', C') is controllable
- Eigenvalues of A' C'L' can be arbitrarily placed
- Note: All the state feedback design procedure can be used to design state estimator, and some of them will be highlighted next

25

#### **Full Dimensional State Estimator, SISO Case**

- **Theorem.** If a SISO system is observable, then it can be transformed, by an equivalent transformation, to an observable canonical form
  - Observable Canonical Form:

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 0 & 0 & \dots & -\alpha_{n} \\ 1 & 0 & 0 & \dots & -\alpha_{n-1} \\ 0 & 1 & 0 & \dots & -\alpha_{n-2} \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & \dots & 1 & -\alpha_{1} \end{bmatrix} \mathbf{x} + \begin{bmatrix} \beta_{n} \\ \beta_{n-1} \\ \beta_{n-2} \\ \vdots \\ \beta_{1} \end{bmatrix} \mathbf{u}; \quad \mathbf{y} = \begin{bmatrix} 0 & 0 & 0 & \dots & 1 \end{bmatrix} \mathbf{x} + d\mathbf{u}$$
$$\mathbf{c}(\mathbf{s}\mathbf{I} - \mathbf{A})^{-1}\mathbf{b} + \mathbf{d} = \frac{\beta_{1}\mathbf{s}^{n-1} + \beta_{2}\mathbf{s}^{n-2} + \dots + \beta_{n}}{\mathbf{s}^{n} + \alpha_{1}\mathbf{s}^{n-1} + \dots + \alpha_{n-1}\mathbf{s} + \alpha_{n}} + \mathbf{d}$$

The transformation matrix Q is formed as follows:

$$q_{n} \equiv c'$$

$$q_{n-1} \equiv A'q_{n} + \alpha_{1}c' \equiv A'c' + \alpha_{1}c'$$

$$q_{n-i} \equiv A'q_{n-i+1} + \alpha_{i}c' \equiv (A')^{i-1} + \alpha_{1}(A')^{i-2} + \dots + \alpha_{i-1}c'$$

$$q_{1} \equiv A'q_{2} + \alpha_{n-1}c' \equiv (A')^{n-1} + \alpha_{1}(A')^{n-2} + \dots + \alpha_{n-1}c'$$
For example, with n=4
$$Q = \left[ (A')^{3}c' \quad (A')^{2}c' \quad A'c' \quad c' \right] \begin{bmatrix} 1 & 0 & 0 & 0 \\ \alpha_{1} & 1 & 0 & 0 \\ \alpha_{2} & \alpha_{1} & 1 & 0 \\ \alpha_{3} & \alpha_{2} & \alpha_{1} & 1 \end{bmatrix}$$

- With  $P \equiv Q'$ , then (PAP<sup>-1</sup>, Pb, cP<sup>-1</sup>) is in the observable canonical form.

27

**Example** 
$$\dot{\mathbf{x}} = \begin{bmatrix} 1 & 2 & 0 \\ 3 & -1 & 1 \\ 0 & 2 & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 2 \\ 1 \\ 1 \end{bmatrix} \mathbf{u}; \quad \mathbf{y} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix} \mathbf{x}$$

- Is the system observable?

$$G^{\circ} = \begin{bmatrix} C \\ CA \\ CA^{2} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 \\ 0 & 2 & 0 \\ 6 & -2 & 2 \end{bmatrix}$$
$$|G^{\circ}| = -12 \neq 0 \quad \sim \text{Observable}$$
$$\Delta(\lambda) = \begin{vmatrix} \lambda - 1 & -2 & 0 \\ -3 & \lambda + 1 & -1 \\ 0 & -2 & \lambda \end{vmatrix} = \lambda (\lambda^{2} - 1) - 2(\lambda - 1) - 6\lambda$$
$$= \lambda^{3} + 0\lambda^{2} - 9\lambda + 2 \qquad \sim \alpha_{1} = 0, \ \alpha_{2} = -9, \ \alpha_{3} = 2$$

$$q_{3} = c' = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}; \qquad q_{2} = A'q_{3} + \alpha_{1}c' = \begin{bmatrix} 0 \\ 2 \\ 0 \end{bmatrix}$$
$$q_{1} = A'q_{2} + \alpha_{2}c' = \begin{bmatrix} 6 \\ -2 \\ 2 \end{bmatrix} - \begin{bmatrix} 0 \\ 0 \\ 9 \end{bmatrix} = \begin{bmatrix} -6 \\ -2 \\ -7 \end{bmatrix}$$
$$Q = \begin{bmatrix} 6 & 0 & 0 \\ -2 & 2 & 0 \\ -7 & 0 & 1 \end{bmatrix}; \qquad P \equiv Q' = \begin{bmatrix} 6 & -2 & -7 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix}; \qquad \overline{A} = \begin{bmatrix} 0 & 0 & -2 \\ 1 & 0 & 9 \\ 0 & 1 & 0 \end{bmatrix}$$
$$\overline{c} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix}; \qquad \overline{b} = Pb = \begin{bmatrix} 3 \\ 2 \\ 1 \end{bmatrix}$$

Now, how to select L?
 Consider the equivalent system in Observable Canonical Form:

$$\begin{split} \dot{\overline{x}} &= \begin{bmatrix} 0 & 0 & 0 & \dots & -\alpha_{n} \\ 1 & 0 & 0 & \dots & -\alpha_{n-1} \\ 0 & 1 & 0 & \dots & -\alpha_{n-2} \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & \dots & 1 & -\alpha_{1} \end{bmatrix} \overline{x} + \begin{bmatrix} \beta_{n} \\ \beta_{n-1} \\ \beta_{n-2} \\ \vdots \\ \beta_{1} \end{bmatrix} u; \quad y = \begin{bmatrix} 0 & 0 & 0 & \dots & 1 \end{bmatrix} \overline{x} \\ \dot{\overline{e}} &= (\overline{A} - \overline{L}\overline{c})\overline{e} \\ \overline{A} - \overline{L}\overline{c} &= \begin{bmatrix} 0 & 0 & 0 & \dots & -\alpha_{n} \\ 1 & 0 & 0 & \dots & -\alpha_{n-1} \\ 0 & 1 & 0 & \dots & -\alpha_{n-2} \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & \dots & 1 & -\alpha_{1} \end{bmatrix} - \begin{bmatrix} \overline{l}_{1} \\ \overline{l}_{2} \\ \overline{l}_{3} \\ \vdots \\ \overline{l}_{n} \end{bmatrix} \begin{bmatrix} 0 & 0 & 0 & \dots & 1 \end{bmatrix}$$

$$\overline{\mathbf{A}} - \overline{\mathbf{L}}\overline{\mathbf{c}} = \begin{bmatrix} 0 & 0 & 0 & \dots & -(\alpha_{n} + \overline{l}_{1}) \\ 1 & 0 & 0 & \dots & -(\alpha_{n-1} + \overline{l}_{2}) \\ 0 & 1 & 0 & \dots & -(\alpha_{n-2} + \overline{l}_{3}) \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & \dots & 1 & -(\alpha_{1} + \overline{l}_{n}) \end{bmatrix}$$

- The characteristic polynomial:

$$\Delta_{d}(s) = s^{n} + (\alpha_{1} + \bar{l}_{n})s^{n-1} + ... + (\alpha_{n-1} + \bar{l}_{2})s + (\alpha_{n} + \bar{l}_{1})$$

– Suppose that the desired estimator poles are  $\{\hat{\lambda}_i\}_{i=1}^n$  and the desired characteristic function is

$$\Delta_{d}(s) = s^{n} + \hat{\alpha}_{1}s^{n-1} + \ldots + \hat{\alpha}_{n-1}s + \hat{\alpha}_{n}$$

Then it is clear that

$$\bar{l}_1 = \hat{\alpha}_n - \alpha_n; \bar{l}_2 = \hat{\alpha}_{n-1} - \alpha_{n-1}; ... \bar{l}_{n-1} = \hat{\alpha}_2 - \alpha_2; \bar{l}_n = \hat{\alpha}_1 - \alpha_1$$

$$\overline{L} = \begin{bmatrix} \hat{\alpha}_n - \alpha_n \\ \hat{\alpha}_{n-1} - \alpha_{n-1} \\ \vdots \\ \hat{\alpha}_1 - \alpha_1 \end{bmatrix}$$

- Finally, Let  $L = P^{-1} \overline{L}$ , then

$$A - LC = P^{-1}\overline{A}P - P^{-1}\overline{L}\overline{C}P = P^{-1}(\overline{A} - \overline{L}\overline{C})P$$

A-LC also has the desired eigenvalues.

 L is uniquely determined by the desired eigenvalues of the estimator.

# Procedure for designing the estimator gain

Step 1. Choose the desired eigenvalue set  $\{\lambda_i, i=1,2,...n\}$  of A-Lc and obtain the coefficients of

$$\Delta_{d}(s) = (s - \lambda_{1})(s - \lambda_{2}) \cdots (s - \lambda^{n}) = s^{n} + \overline{\alpha}_{1}s^{n-1} + \cdots + \overline{\alpha}_{n-1}s + \overline{\alpha}_{n}$$

Step 2. Compute the characteristic polynomial of A

$$\Delta(s) = s^n + \alpha_1 s^{n-1} + \dots + \alpha_{n-1} s + \alpha_n$$

and the transformation matrix, e.g., for n = 4

$$Q = \begin{bmatrix} (A')^{3} c' & (A')^{2} c' & A'c' & c' \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 & 0 \\ \alpha_{1} & 1 & 0 & 0 \\ \alpha_{2} & \alpha_{1} & 1 & 0 \\ \alpha_{3} & \alpha_{2} & \alpha_{1} & 1 \end{bmatrix}, P = Q'$$

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Then 
$$\overline{A} = PAP^{-1} = \begin{bmatrix} 0 & 0 & 0 & -\alpha_4 \\ 1 & 0 & 0 & -\alpha_3 \\ 0 & 1 & 0 & -\alpha_2 \\ 0 & 0 & 1 & -\alpha_1 \end{bmatrix}$$
,  $\overline{c} = CP^{-1} = \begin{bmatrix} 0 & 0 & 0 & 1 \end{bmatrix}$ ,  
 $\overline{A} \cdot \overline{bk} = \begin{bmatrix} 0 & 0 & 0 & -\alpha_4 - \overline{l_1} \\ 1 & 0 & 0 & -\alpha_3 - \overline{l_2} \\ 0 & 1 & 0 & -\alpha_2 - \overline{l_3} \\ 0 & 0 & 1 & -\alpha_1 - \overline{l_4} \end{bmatrix}$   
Step 3: Choose  $\overline{l_i} = \overline{\alpha_i} - \alpha_i$   
Then  $\overline{A} \cdot \overline{Lc} = \begin{bmatrix} 0 & 0 & 0 & -\overline{\alpha_4} \\ 1 & 0 & 0 & -\overline{\alpha_3} \\ 0 & 1 & 0 & -\overline{\alpha_2} \\ 0 & 0 & 1 & -\overline{\alpha_1} \end{bmatrix}$   
Step 4: Compute  $L = P^{-1}\overline{L}$ .

then A - Lc = P<sup>-1</sup>(A –  $\overline{L}\overline{c})P$  has the desired eigenvalues  $\{\lambda_i, i=1,2,...,n\}$ 

# Example (Continued)

$$\dot{\mathbf{x}} = \begin{bmatrix} 1 & 2 & 0 \\ 3 & -1 & 1 \\ 0 & 2 & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 2 \\ 1 \\ 1 \end{bmatrix} \mathbf{u}; \quad \mathbf{y} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix} \mathbf{x}$$

- Find L s.t. the estimator poles are all at -3, -3, -10
- As discussed earlier, the system is observable, and

$$\Delta(\lambda) = \lambda^3 + 0\lambda^2 - 9\lambda + 2 \quad \sim \alpha_1 = 0, \ \alpha_2 = -9, \ \alpha_3 = 2$$

- The desired estimator characteristic function  $A_{1}(x) = (x + 3)^{2}(x + 10) = x^{3} + 16x^{2} + 60x + 00$ 

$$\Delta_{d}(s) = (s+3)^{2}(s+10) = s^{3} + 16s^{2} + 69s + 90$$

$$\Rightarrow \overline{\alpha}_1 = 16, \, \overline{\alpha}_2 = 69, \, \overline{\alpha}_3 = 90; \qquad \overline{L} = \begin{bmatrix} \overline{\alpha}_3 - \alpha_3 \\ \overline{\alpha}_2 - \alpha_2 \\ \overline{\alpha}_1 - \alpha_1 \end{bmatrix} = \begin{bmatrix} 88 \\ 78 \\ 16 \end{bmatrix}$$

2	5
2	з

- From the previous example,  $\begin{bmatrix} 1 & 1 & 7 \end{bmatrix}$ 

$$P = Q^{T} = \begin{bmatrix} 6 & -2 & -7 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix}; P^{-1} = \begin{bmatrix} \frac{1}{6} & \frac{1}{6} & \frac{7}{6} \\ 0 & \frac{1}{2} & 0 \\ 0 & 0 & 1 \end{bmatrix}$$
$$L = P^{-1}\overline{L} = \begin{bmatrix} \frac{1}{6} & \frac{1}{6} & \frac{7}{6} \\ 0 & \frac{1}{2} & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 88 \\ 78 \\ 16 \end{bmatrix} = \begin{bmatrix} 139/3 \\ 39 \\ 16 \end{bmatrix}$$

- Verification:

$$A - Lc = \begin{bmatrix} 1 & 2 & 0 \\ 3 & -1 & 1 \\ 0 & 2 & 0 \end{bmatrix} - \begin{bmatrix} 139/3 \\ 39 \\ 16 \end{bmatrix} \begin{bmatrix} 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} 1 & 2 & -139/3 \\ 3 & -1 & -38 \\ 0 & 2 & -16 \end{bmatrix}$$

$$\begin{aligned} |\lambda - (A - Lc)| &= \begin{vmatrix} \lambda - 1 & -2 & \frac{139}{3} \\ -3 & \lambda + 1 & 38 \\ 0 & -2 & \lambda + 16 \end{vmatrix} \\ &= (\lambda^2 - 1)(\lambda + 16) + 278 + 76(\lambda - 1) - 6(\lambda + 16) \\ &= (\lambda^3 + 16\lambda^2 - \lambda - 16) + 278 + (76\lambda - 76) - (6\lambda + 96) \\ &= \lambda^3 + 16\lambda^2 + 69\lambda + 90 \quad \sim \text{As desired} \end{aligned}$$

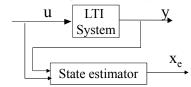
37

Today's topics:

- Other state-feedback problems
  - Regulation and tracking
  - Robust tracking and disturbance rejection
  - Stabilization of uncontrollable systems
- Full dimensional estimator
  - SISO case via observable canonical form
  - MIMO case by solving matrix equation

# **General MIMO State Estimator**

- Dual to MIMO state feedback ~ Methods discussed earlier has a counterpart here
  - We will discuss the method based on matrix equation, assuming that the system is observable



- A full-dimensional state estimator with D = 0:

$$\dot{\mathbf{x}}_e = \mathbf{A}\mathbf{x}_e + \mathbf{B}\mathbf{u} + \mathbf{L}(\mathbf{y} - \mathbf{y}_e), \quad \mathbf{y}_e = \mathbf{C}\mathbf{x}_e$$

Let  $e=x-x_e$ , the error dynamics:

$$\dot{\mathbf{e}} = (\mathbf{A} - \mathbf{L}\mathbf{C})\mathbf{e}$$

Similar to the state feedback design, the objective is to pick an observer gain L such that A-LC is equivalent to a certain F which has the desired eigenvalues, i.e.,

 $A - LC = T^{-1}FT$  for a nonsingular T

The procedure: Given A,C and F. Pick  $L_0$ , solve  $TA - FT = L_0C$  for T. If T is nonsingular, let  $L=T^{-1}L_0$ . Then A – LC has the desired eigenvalues.

About the solution to  $TA - FT = L_0C$ , we have:

- The matrix equation has a unique solution iff A and F have no common eigenvalues;
- The solution T is nonsingular only if (A,C) is observable and (F,L<sub>0</sub>) is controllable;
- In case that C has one row, T is nonsingular if and only if (A,C) is observable and (F,L<sub>0</sub>) is controllable.

The above result can be derived from the results for state feedback design from duality: taking transpose of the equation, we obtain

A'T'- T'F' = C'L<sub>0</sub>' as compared with A T - T F = B K<sub>0</sub> for the state feedback design<sub>41</sub>

## Example

$$\dot{\mathbf{x}} = \begin{bmatrix} 1 & 2 & 0 \\ 3 & -1 & 1 \\ 0 & 2 & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 2 \\ 1 \\ 1 \end{bmatrix} \mathbf{u}; \quad \mathbf{y} = \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 1 \end{bmatrix} \mathbf{x}$$

Find T s.t. the estimator poles are at -5, -5, -10 – Check observability:

$$G^{\circ} = \begin{bmatrix} C \\ CA \\ CA^{2} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 1 \\ 0 & 2 & 0 \\ 3 & 1 & 1 \\ x & x & x \\ x & x & x \end{bmatrix}$$

 $\sim$  Observable

$$- \text{ Select F: } F = \begin{bmatrix} -5 & 0 & 0 \\ 0 & -5 & 0 \\ 0 & 0 & -10 \end{bmatrix}$$

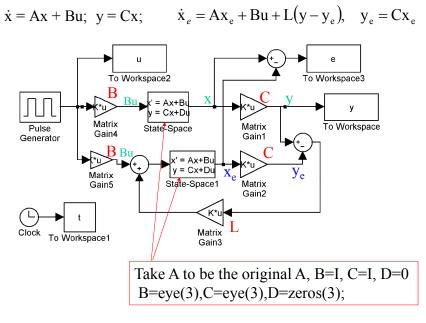
$$- \text{ Select } L_0: \quad L_0 = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 2 \end{bmatrix} \implies \{F, L_0\} \text{ controllable}$$

$$- \text{ Solve TA-FT} = L_0C \text{ with matlab: } T = \text{lyap}(-F, A, -L0*C)$$

$$T = \begin{bmatrix} 0.1250 & -0.2500 & 0.2500 \\ -0.1875 & 0.3750 & 0.1250 \\ -0.0602 & 0.2206 & 0.1779 \end{bmatrix}, \quad L = T^{-1}L_0 = \begin{bmatrix} -3.5 & 24 \\ -3.75 & 15 \\ 3 & 2 \end{bmatrix}$$

$$- \text{ Verify: } \text{ eig}(A - LC) = \{-5, -5, -10\}$$

$$- \text{ If we pick: } L_0 = \begin{bmatrix} 1 & 0 \\ -1 & 1 \\ 1 & 0 \end{bmatrix} \Rightarrow L = \begin{bmatrix} -117 & 24 \\ -60.5 & 15 \\ 3 & 2 \end{bmatrix}$$



Explanation:

$$\dot{u}$$
  $\dot{x} = Ax + Bu$   
 $y = Cx$ 

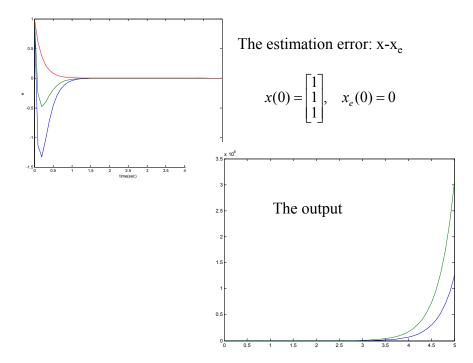
Can be equivalently realized with

$$u \rightarrow B \rightarrow v \qquad \dot{x} = Ax + v \qquad y_1 \rightarrow C \rightarrow y$$
$$y_1 = x \qquad v = Bu, \ y = Cy_1 = Cx$$

The purpose of doing this is to get  $\boldsymbol{x}$  and  $\boldsymbol{x}_e$ 

$$\begin{array}{c} \underbrace{u}_{B} \underbrace{v_{e}}_{e} = Ax_{e} + v_{e} \underbrace{y_{e1}}_{C} \underbrace{y_{e}}_{e} \\ y_{e1} = x_{e} \\ L(y-y_{e}) \end{array} v_{e} = Bu+L(y-y_{e}), y_{e} = Cy_{e1} = Cx_{e}$$

$$\begin{array}{c} 45 \\ 45 \end{array}$$



#### **Observer for discrete-time system**

The original system  $x(k+1) = Ax(k) + Bu(k); \ y(k) = Cx(k)$ The duplicated system (estimator)  $x_e(k+1) = Ax_e(k) + Bu(k) + L(y(k) - y_e(k)), \quad y_e(k) = Cx_e(k)$ Define the observer error,  $e(k) := x(k) - x_e(k)$  $e(k+1) = (Ax(k) + Bu(k)) - [(A - LC)x_e(k) + LCx(k) + Bu(k)]$  = (A - LC)e(k) e(k+1) = (A - LC)e(k)

If the eigenvalues of (A - LC) are inside the unit disk, |λ<sub>i</sub>(A)| < 1, any error will converge to 0, x(k) − x<sub>e</sub>(k) → 0.

Use same algorithms to assign the eigenvalues of A-LC.

47

Problem set #11 2. For the system  $\dot{x} = \begin{bmatrix} 1 & 2 & 0 \\ -1 & -1 & -1 \\ 0 & 2 & -1 \end{bmatrix} x + \begin{bmatrix} 2 \\ 0 \\ 1 \end{bmatrix} u; \quad y = \begin{bmatrix} 1 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix} x$ Design an observer to estimate the state x. The poles of the observer are -4+j4, -4-j4 and -8. Simulate the system from t=0 to t=10 with x(0)=[1 1 1]', x\_e(0)=[0 0 0]' and u=0. Plot the state x(t) in one figure and e = x(t)-x\_e(t) in another figure. Also print the simulink model. Today's topics:

- Other state-feedback problems
  - Regulation and tracking
  - Robust tracking and disturbance rejection
  - Stabilization of uncontrollable systems
- Full dimensional estimator
  - SISO case via observable canonical form
  - MIMO case by solving matrix equation

Next Time:

- Reduced-order estimator
- Connection of state-feedback with state estimation
- LQR optimal control

Problem set #11: see slides 16, 48