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Buckling of Rigid Frames – I

Prof. Tzuyang Yu

Structural Engineering Research Group (SERG) Department of Civil and Environmental Engineering University of Massachusetts Lowell Lowell, Massachusetts

SERG

Outline

- Effect of geometrical imperfection
 - $\ensuremath{\,\mathsf{P}}\xspace{-}\delta$ and $\ensuremath{\mathsf{P}}\xspace{-}\Delta$ effects
- Load-deflection behavior of frames
- Analysis approaches
 - D.E. method
 - Slope-deflection method
 - Matrix stiffness method
- Slope-deflection method
- Elastic critical loads D.E. method
 - Non-sway case
 - Sway case
- Summary
- References



• Moment distribution





(Source: J.E. Steel, AZ)







- Effects of geometric imperfection
 - P- δ and P- Δ effects

Load-deflection behavior of frames

- Elastic buckling load, P_{cr}
- Plastic collapse load, P_p
- Actual failure load, P_f

- Analysis approaches
 - D.E. method
 - Force based
 - Second-order and fourth-order
 - Slope-deflection method
 - Displacement based
 - Matrix form
 - Matrix stiffness method
 - Force based
 - Matrix form

- Slope deflection method is a displacement-based analysis for indeterminate structures
- Unknown displacements are first written in terms of the loads by using load-displacement relationships; then these equations are solved for the displacements.
- Once the displacements are obtained, unknown loads are determined from the compatibility equations using load-displacement relationships.
- Nodes: Specified points on the structure that undergo displacements (and rotations).
- Degrees of Freedom (DOF): These displacements (and rotations) are referred to as degrees of freedom.

To clarify these concepts we will consider some examples, beginning with the **beam in Fig. 1(a).** Here any load P applied to the beam will cause node A only to rotate (neglecting axial deformation), while node B is completely restricted from moving. Hence the beam has only one degree of freedom, θ_A .



Fig. 1 (a)

The beam in Fig. 1(b) has node at A, B, and C, and so has four degrees of freedom, designed by the rotational displacements θ_A , θ_B , θ_C , and the vertical displacement Δ_C .



Fig. 1 (b)

Consider now the frame in Fig. 1(c). Again, if we neglect axial deformation of the members, an arbitrary loading P applied to the frame can cause nodes B and C to rotate nodes can be displaced horizontally by an equal amount. The frame therefore has three degrees of freedom, θ_A , θ_B , Δ_B .



Fig. 1 (c)

Consider portion AB of a continuous beam, shown below, subjected to a distributed load w(x) per unit length and a support settlement of Δ at B; EI of the beam is constant.





Governing equation –

$$\begin{split} M_{AB} &= \mathrm{FEM}_{AB} + M_{AB}' + M_{AB}'' + M_{AB}''' \\ &= \mathrm{FEM}_{AB} + \frac{4EI\theta_A}{L} + \frac{2EI\theta_B}{L} - \frac{6EI\Delta}{L^2} \\ &= \frac{2EI}{L} \left(2\theta_A + \theta_B - \frac{3\Delta}{L} \right) + \mathrm{FEM}_{AB} \end{split}$$

$$\begin{split} M_{BA} &= \mathrm{FEM}_{BA} + M_{BA}' + M_{BA}'' + M_{BA}''' \\ &= \mathrm{FEM}_{BA} + \frac{2EI\theta_A}{L} + \frac{4EI\theta_B}{L} - \frac{6EI\Delta}{L^2} \\ &= \frac{2EI}{L} \left(\theta_A + 2\theta_B - \frac{3\Delta}{L} \right) + \mathrm{FEM}_{BA} \end{split}$$





 $M_{AB} = [(FEM)_{AB} - (FEM)_{BA}/2] + (3EI\theta_A)/L - (3EI\Delta)/L^2$ Modified FEM at end A

• Elastic critical load – Differential equation approach

- Non-sway case
- 1. Governing equations
 - 1.1 For the beam

1.2 For the column

2. Displacement solution & boundary conditions

• Elastic critical load – Differential equation approach

- Non-sway case
- 3. Compatibility condition

4. Characteristic equation \rightarrow Critical load, P_{cr}

• Elastic critical load – Differential equation approach

- Sway case
- 1. Governing equations
 - 1.1 For the beam

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4. Characteristic equation \rightarrow Critical load, P_{cr}

- Real structures, such as buildings, behave like frames. Boundary conditions and joint conditions become critical in determining the critical load of the structures.
- Geometric imperfection plays a key role in the critical load.
- Secondary effects (P-δ and P-Δ effects) will make the elastic load-deflection behavior become nonlinear (geometric nonlinearity)

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